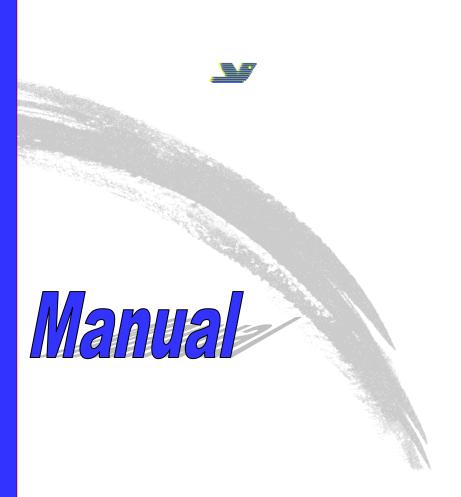
Karma / Karma SP





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!!! WARNING

This is not a training manual. It is extremely dangerous to yourself and others to attempt to fly this or any paraglide without first completing a flying course given by a qualified instructor.

Apco Aviation's gliders are carefully manufactured and inspected by the factory. Please use the glider only as described in this manual. Do not make any changes to the glider. AS WITH ANY SPORT - WITHOUT TAKING THE APPROPRIATE PRECAUTIONS, PARAGLIDING CAN BE DANGEROUS.





1 KARMA TECHNICAL DATA

Size		X-Small	Small		Medium	Large
Cells 38		38	39		41	43
Area m ²		25.6	26.6		28.6	30.6
Area (projected) m ²		22.6	23.6		25.6	27.6
Span (incl. Stabiliser) m		11.06	11.41		12.12	12.83
Span (projected) m		8.79	9.14		9.85	10.56
Aspect Ratio		4.8	4.9		5.1	5.4
Aspect Ratio (projected)		3.4	3.5		3.8	4.0
Pilot Weight, Kg (all up)		50-75	60-90		80-110	100-130
Weight of Canopy Kg		5.6	6		6.3	6.6
Root Cord m		2.84	2.84		2.84	2.84
Tip Cord m		0.48	0.48		0.48	0.48
V-min (km/h)	V-min (km/h)		20		20	20
V-trim (km/h)	V-trim (km/h)		36		36	36
V-max (km/h)		45+	45+		45+	45+
Min Sink (at optimum wing loading	ıg)	1.1	1.1		1.1	1.1
Length of Lines on B m		7.4	7.7		8.2	8.7
Total length of line used m		374	412		448	473
		LINES				
	N	Material	Diameter		Strengt	h
Тор	Sup	oer Aramid	1.2 mm		110 kg	7
Bottom A3,5;B3,5;C1,3,5;D1,3	Super Aramid		1.8 mm	220 kg		7
Bottom A1;B1	Super Aramid		1.9 mm	320 kg		7
Brake Cascades	Dyneema		1.1 mm	95 kg		
Steering Line Polyester		2.0 mm		85 kg		
	FABRIC					
Sail Clot			42gr/m ² "Zero Porosity" Ripstop Nylon			
Warranty					3 Years / 250 hours	3

2 KARMA CERTIFICATION DATA

KARMA EXTRA-SMALL	DHV 1 trim & accel
KARMA SMALL	DHV 1 trim & accel
KARMA MEDIUM	DHV 1 trim & accel
KARMA LARGE	DHV 1 trim & accel





3 KARMA SP TECHNICAL DATA

Size		X-Small	Small	Medium	Large
Cells		38	39	41	43
Area m ²		25.6	26.6	28.6	30.6
Area (projected) m ²		22.6	23.6	25.6	27.6
Span (incl. Stabiliser) m		11.06	11.41	12.12	12.83
Span (projected) m		8.79	9.14	9.85	10.56
Aspect Ratio		4.8	4.9	5.1	5.4
Aspect Ratio (projected)		3.4	3.5	3.8	4.0
Pilot Weight, Kg (all up)	Continue Co	50-75	60-90	80-110	100-130
Weight of Canopy Kg		5.2	5.4	5.7	6.0
Root Cord m		2.84	2.84	2.84	2.84
Tip Cord m		0.48	0.48	0.48	0.48
V-min (km/h)		20	20	20	20
V-trim (km/h)	4.97	37	37	37	37
V-max (km/h)	V-max (km/h)		46+	46+	46+
Min Sink (at optimum wing loadi	ng)	1.1	1.1	1.1	1.1
Length of Lines on B m		7.4	7.7	8.2	8.7
Total length of line used m		374	412	448	473
		LINES			
	N	Material	Diameter	Streng	th
Top a; b; St	D	yneema	1.1 mm	95 kg	5
Top c; d; Lower St	Super Aramid		1.2 mm	110 kg	
Bottom C1,3,5;D1,3;Mid a,b	Super Aramid		1.5 mm	150 kg	
Bottom A3,5;B3,5	Super Aramid		1.8 mm	230 kg	
Bottom A1;B1 Super Aramid		oer Aramid	1.9 mm	320 kg	g
Brake Cascades	D	yneema	1.1 mm	95 kg	
Steering Line	Р	olyester	2.0 mm	85 kg	STERRING, WA

	FABRIC		
Sail Clo	th	42g	r/m² "Zero Porosity" Ripstop Nylon
Warran	ty		3 Years / 250 hours
	*		

4 KARMA SP CERTIFICATION DATA

KARMA SP X-SMALL	
KARMA SP SMALL	EN-B
KARMA SP MEDIUM	EN-B
KARMA SP LARGE	EN-B





5 DISCLAIMER OF LIABILITY

Taking into consideration the inherent risk in paragliding, it must be expressly understood that the manufacturer and seller do not assume any responsibility for accidents, losses and direct or indirect damage following the use or misuse of this product.

APCO Aviation Ltd. is engaged in the manufacture and sale of hang gliding, paragliding, motorized Para/hang gliding and emergency parachute equipment.

This equipment should be used under proper conditions and after proper instruction from a qualified instructor. APCO Aviation Ltd. has no control over the use of this equipment and a person using this equipment assumes all risks of damage or injury.

APCO Aviation Ltd. disclaims any liability or responsibility for injuries or damages resulting from the use of this equipment.

The glider is designed to perform in the frame of the required class as certified.

6 CONSTRUCTION

The glider is constructed with a top and bottom surface, connected by ribs. One top and bottom panel, together with the connecting ribs is called a cell. Each cell has an opening on the front lower part. The cells fill with air forcing the panels to take the shape dictated by the airfoil (rib) section. On either side the wing ends in a stabilizer or wing tip, which provides straight-line (Yaw) stability and produces some outward force to keep the spanwise tension. The front part of the ribs is made from Trilam to keep the leading edge shaped at high speeds and in turbulent air. It also improves the launch characteristics of the glider. The line hook-up points are made of Dyneema or Nylon tape.

7 MATERIALS

The glider is made from tear resistant Ripstop Nylon cloth, which is P.U. coated to zero porosity and then siliconized to give the fabric high resistance to the elements. Different cloth is used for the top, bottom and ribs due to their different functions. The lines are made of superaramid covered with a polyester sheath for protection against UV, wear and abrasion. The bottom section of the brake lines is made of polyester because of its better mechanical properties. The karabiners that attach the lines to the risers are made of stainless steel.

8 TRIMMING

All Apco gliders are trimmed for optimum performance combined with unsurpassed safety. It is very important not to re-trim or tamper with any of the lines or risers as this may alter the performance and safety. Trimming of the brake line should be done in accordance with this manual and carefully checked before flying.





All of Apco's gliders are developed with the use of ABS (Automatic Bracing System) type harnesses without cross bracing. We recommend the use of an ABS harness with all our gliders. All certified harnesses can be used with our gliders. For best safety and performance we recommend an Apco harness equipped with a Mayday emergency parachute.

CAUTION:

WE STRICLY RECOMEND NOT TO USE CROSS BRACING STRAPS.

DUE TO APCO GLIDERS ARE DEVELOPED AND TESTED WITHOUT THE USE OF CROSS BRACING. USING AN ABS HARNESS WITH CHEST STRAP SET AT THE SPECIFIED WIDTH (CHECK THE CERTIFICATE STICKER ON YOUR GLIDER) WILL RESULT IN THE HIGHEST PASSIVE SAFETY ON YOUR GLIDER.

10 SPEED SYSTEM

10.1 ASSEMBLY & ADJUSTMENT

Apco gliders are supplied with a speed system as illustrated in option II. The pilot can change the speed system to the traditional "Apco" speed system to use the full accelerator range depending on the pulley arrangement on the harness being used.

10.2 OPTION 1

First attach the harness to the glider. Remove the Chain Link from the end of the accelerator line attached to the speed bar, then thread it through the elasticized ring on the harness, then through harness pulley and then re-attach it to the Chain link with a larkshead knot. Hook the Chain link onto the Chain Link on the riser of the corresponding side. Sit in the harness and have someone hold the riser up in a flying position for you. Adjust the speed bar line by pulling the end through the speed bar tube and moving the knot. The Bar should be about 10 cm (or closer if you have a second step) away from the front of the harness seat. This allows you to easily reach the bar with your foot, and will allow you to use the full range of the speed bar if you extend your legs fully. Do not adjust the speed system too short, as this will cause it to be activated permanently while flying, and could be dangerous. It is possible to fit a second step to the system, if one has trouble using the full range of the system (supplied separately).



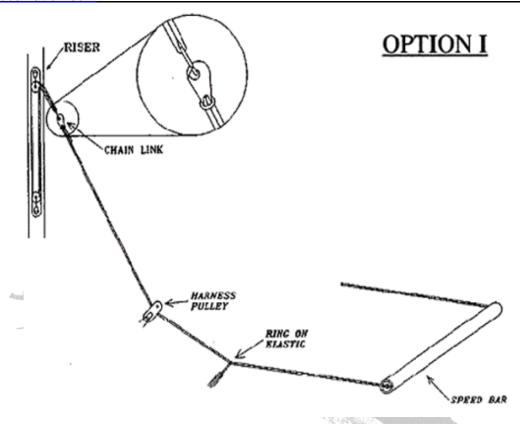
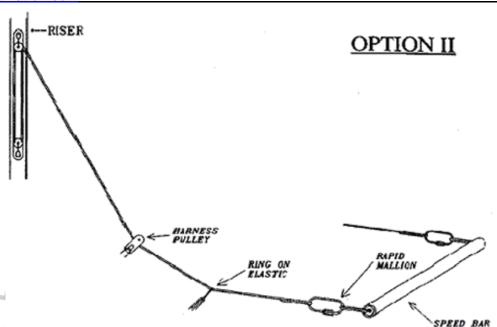


Figure 10-1

10.30PTION 2

First attach the harness to the glider, then thread the accelerator line from the top pulley on the riser, through the pulleys (and the elasticized ring if present) on your harness and then attach it to the supplied speed bar. To adjust the length and activation point of the speed system, sit in the harness and ask someone to hold the riser up in a flying position. By pulling out the end of the line protruding above the upper pulley on the riser and by moving the knot you can adjust the speed system. The Bar should be about 10 cm (or closer if you have a second step) away from the front of the harness seat. This allows you to use the full range of the speed bar if you extend your legs fully. Do not adjust the speed system too short as this will cause the speed system to be activated permanently while flying and could be dangerous. It is possible to fit a second step to the system if the pilot has trouble using the full range of the speed system (second step is supplied separately).





WARNING:

The use of the speed system in turbulent conditions or close to the ground is dangerous. While flying with the accelerator, the glider has a reduced angle of attack and is therefore more susceptible to turbulence and may collapse or partially deflate. Gliders react faster when accelerated and may turn more. The accelerator should immediately be released in this case.

Figure 10-2

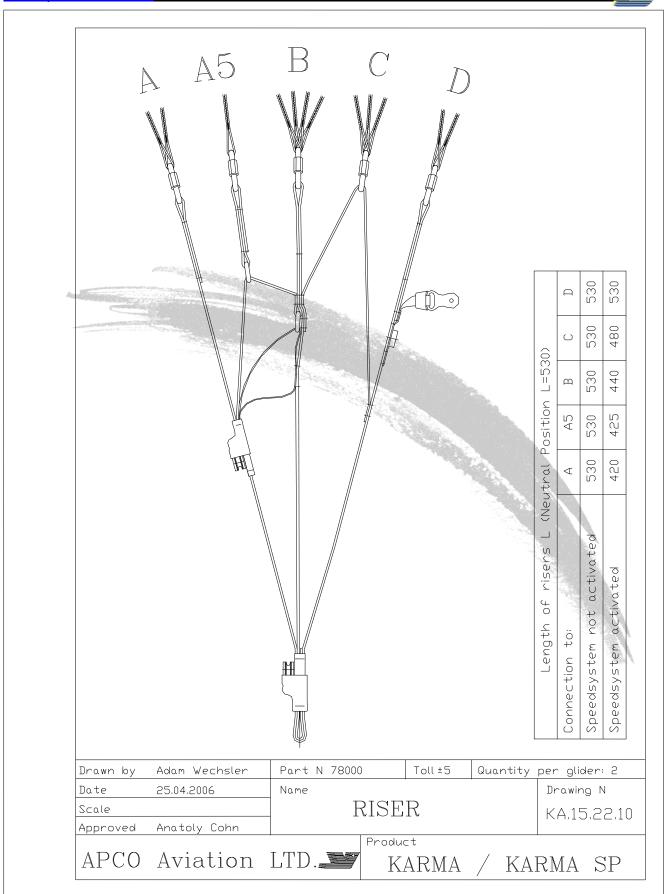
11 EMERGENCY PARACHUTE ATTACHMENT

It is recommended to use a certified rescue parachute when flying. Attaching the rescue parachute should be done in accordance with the recommendation of the harness and reserve parachute manufacturer.

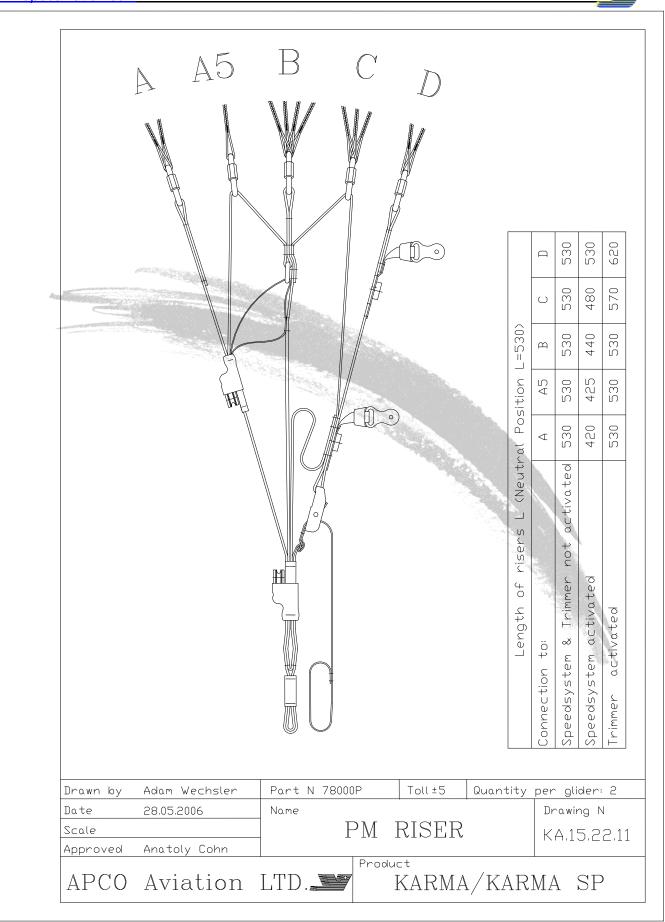
12 RISERS

The KARMA / KARMA SP is supplied with a split A riser. The 1st A-riser attaches to the central two A lines (A1 & A3). The second A-riser is attached to the outermost A line (A5). This is to facilitate Big ears or Tip tucks. At no time should the pilot change the risers or use risers not intended for this specific glider as this will affect the performance and safety of the glider.













13.1 GENERAL

Pilots, please insure that your glider has been test flown and checked by your dealer before taking it into your possession.

13.2 BRAKE SETTING

Before the first flight the pilot/dealer has to take his/her glider and inflate it to check brake length and if needed shorten or lengthen the brake setting to his or her preference. It is important that the brakes are not set too short. If the glider is above your head the brakes should not be pulling the trailing edge down as that means that the brakes are too short. A good setting is to have about 10 cm of slack in the brake from the brake guide on the riser to the activation point of the brakes. (See Diagram below)

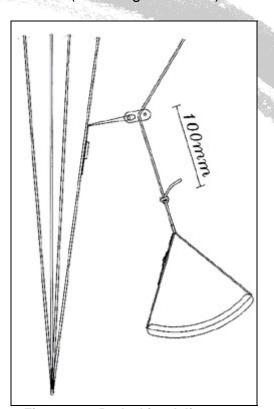


Figure 13-1 Brake Line Adjustment

13.3 FIRST CHECK AND PREFLIGHT INSPECTION

With every new glider, the following points should be checked:

- Connection points between the glider and the harness.
- Check that there are no lines twisted, tangled or knotted.
- Check that the risers and speed-system are hooked up to the harness correctly.





- Damage to lines, webbing and thread on the stitching of the harness and risers.
- That the stainless steel connection links on the risers are not damaged and are fully closed.
- The pulleys of the speed system are free to move and the lines are not twisted.
- The condition of the brake lines, stainless steel ring and the security of the knot attaching the brake handle to the brake line.
- The sewing, condition of the lines and connection of the lines.
- Damage to hook up points on the glider.
- Internal damage to the ribs and diagonal ribs.
- Damage to the top and bottom panels and seams between panels.

13.5 LINE MAINTENANCE

Several groups of suspension lines and one brake line are attached to each riser. The groups are called A, B, C, D and brake lines. The stabilizer lines are connected along with the B-lines. Superaramid lines are known to be sensitive to the influence of the elements. They must be carefully inspected periodically. In his/her own interest, the pilot must observe the following points to ensure maximum performance and safety from the glider.

- Avoid sharp bending and squeezing of lines.
- Take care that people do not step on the lines.
- Do not pull on the lines if they are caught on rocks or vegetation.
- Avoid getting the lines wet. If they do get wet, dry them as soon as possible at room temperature and never store them wet.

IT IS **STRICTLY RECOMMENDED** TO CHANGE THE BOTTOM LINES ON EVERY PARAGLIDER ONCE A YEAR OR EVERY 100 HOURS, WHICH EVER COMES FIRST. THE REST OF THE LINES MUST BE CHECKED YEARLY AND REPLACED IF NECESSARY. THIS RECOMMENDATION IS IN LINE WITH ISRAELI REGULATIONS, BINDING IN ISRAEL. AS AN ALTERNATIVE, WE SUGGEST FOR YOU TO FOLLOW THE REGULATIONS SET BY YOUR NATIONAL AUTHORITIES WITH REGARD TO LINE MAINTENANCE AND REPLACEMENT.

NEVER REPLACE THE LINES WITH DIFFERENT DIAMETER OR TYPE OF LINES AS ALL GLIDERS WERE LOAD TESTED FOR SAFETY IN THEIR ORIGINAL CONFIGURATION. CHANGING LINE DIAMETER/STRENGTHS CAN HAVE FATAL CONSEQUENCES.

Every six months one lower A, B, C and D line must be tested for minimum 50 % of the rated strength. If the line fails under the load test or does not return to its specified length all the corresponding lines must be replaced.

Professional use of gliders: Towing, tandem, schooling and competition flying requires more frequent line inspection and replacement of A, B, C, D and brake lines.





As this is not a training manual we will not try to teach you launching techniques. We will only briefly go through the different launch techniques to help you get the most out of your glider.

14.1 LAYOUT

Pre-flight check should be done before every flight.

Spread the glider on the ground. Spread the lines, dividing them into eight groups A, B, C, D and brake lines left and right. Make sure the lines are free and not twisted or knotted. Make sure all the lines are on top of the glider and that there are none caught on vegetation or rocks under the glider. Lay out the glider in a horseshoe shape. This method insures that all the lines are equally tensioned on launch, and results in an even inflation. The Mylar rib reinforcements will keep the leading edge open for easy inflation. The most common reason for a bad launch is a bad layout!

14.2 ALPINE LAUNCH OR FORWARD LAUNCH

The KARMA / KARMA SP have very good launch behaviour in no wind conditions. For the best results we recommend the use of the following techniques: Lay out the glider and position yourself in the centre of the wing with the lines almost tight. With a positive and constant force inflate the wing holding only the A-risers, and smoothly increase your running speed. The wing will quickly inflate and settle above the pilots head without the tendency to stick behind, the pilot might have to pull some brake to stop the wing from overshooting the pilot on an aggressive run. After you leave the A-risers, apply about 15% brakes and the KARMA / KARMA SP will gently lift you off the ground.

14.3 STRONG WIND AND REVERSE LAUNCH

The KARMA / KARMA SP have a lot of lifting power and care should be taken in strong wind. It is advisable to have an assistant hold the pilot when attempting a strong wind launch. It also helps if you walk towards the canopy and leave the A-riser just before the glider gets above your head. Then pull a bit of brake to stop the wing from overshooting, but not too much as the glider might pull you off your feet too early. The assistant should let the pilot walk in under the wing on inflation rather than resist the inflation; this reduces the tendency of the glider to lift the pilot prematurely.

14.4 TOW OR WINCH LAUNCHING

All APCO gliders are well suited for winching and have no bad tendencies on the winch. With towing it is important to have the wing above your head on launch and not to try and force a stalled wing into the air. This is especially important if the winch operator is using high tension on the winch. Very little brake if any need be applied on launch and during the winch. Directional changes can be made with weight shift rather than brakes. While on tow, the brake pressure will be higher and more input may be needed to make corrections than in normal flight.

For all our gliders we recommend using tow accelerating system. There are different types existing on the market. Please check with your dealer or tow operator for the recommended tow accelerator system. Use of it will eliminate any chance for accidental stalling on tow.



15 FLIGHT TECHNIQUES

15.1 FLYING SPEED

Indicated trim speed is dependant on the amount of brake the pilot is using, wing loading, altitude above sea level and the accuracy and make of speed probe. The speeds recorded were at optimum wing loading at sea level using a Skywatch Pro; thus there could be a slight variation in speed range numbers that a pilot records.

NOTE:

The speeds indicated in the technical data sheet are the correct speeds measured at sea level using a calibrated instrument flying at optimum wing loading. Speed readings in the flight reports could differ as this was measured during testing using various instruments and is an indication of the difference between trim, stall and top speed. The speed range

- With 0% brake the KARMA / KARMA SP will fly at 36/37 km/h with a sink rate of 1.2m/s.
- At 25% brake the glider will fly at 31km/h with minimum sink rate 1.1 m/s.
- The best glide angle is achieved with 0% brakes.

will be the same but the actual numbers may differ.

• With 80% brake the glider will fly at about 21km/h and will be close to the stall point 20km/h.

CAUTION:

APART FROM WHEN FLARING AT LANDING THERE SHOULD BE NO REASON TO FLY WITH 70% TO 100% BRAKE. THE SINK RATE OF THE GLIDER WILL BE EXCESSIVE AND THERE WILL BE A POSSIBILITY OF ENTERING A DEEPSTALL OR FULLSTALL SITUATION. THERE IS ALSO THE RISK OF GOING NEGATIVE OR ENTERING A SPIN WHEN ATTEMPTING TO TURN THE GLIDER NEAR THE STALL SPEED.

WARNING:

The use of the speed system in turbulent conditions or close to the ground is dangerous. While flying with the accelerator, the glider has a reduced angle of attack and is therefore more susceptible to turbulence and may collapse or partially deflate. Gliders react faster when accelerated and may turn more. The accelerator should immediately be released in this case.

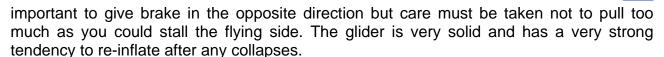
15.2 THERMAL FLYING

The KARMA / KARMA SP have excellent thermaling capacity and will be a pleasure even when you are in a big gaggle or just having fun on a long XC flight. The glider has high internal pressure and needs very little pilot input even in very turbulent conditions. In light lift it is advised to make flat turns to keep the glider from banking too much and avoid increasing the sink rate. In strong lift conditions it is most effective to have small turns in the core with relatively high bank. For the best climb rate in ridge lift we recommend using about 15 to 20% brake.

15.3 ASYMMETRIC COLLAPSE

If one side of the glider partially folds or collapses it is important to keep your flying direction by applying some brake on the opposite side. The wing should re-inflate on its own without any input from the pilot. To help re-inflation it is possible to pull some brake on the collapsed side and release immediately. In the event of a big deflation, i.e. 70%, it is





15.4 CRAVAT

In case a cravat should occur from an asymmetric collapse or other manoeuvres, it is important to keep your flying direction by applying some brake on the opposite side and then it can usually be opened by pulling down on the stabilo line of the affected side while countering the turn with the opposite brake and weight shift. It also helps sometimes to pull Big Ears to release the tension on the affected lines, or a combination of the above techniques, i.e. pulling on the stabilo after pulling Big Ears.

15.5 FRONT STALL OR SYMMETRIC COLLAPSE

In the event of a front stall the glider will normally re-inflate on its own immediately without any change of direction. To speed up re-inflation briefly apply 30-40% brake (to pump open the leading edge). Do not hold the brakes down permanently to avoid an unwanted stall.

15.6 B-STALL

The KARMA / KARMA SP have a very clean stable B stall. To enter the B stall the pilot has to pull the first 20-cm slowly until the glider loses forward speed and starts to descend vertically. Then the pilot can pull more on the B until he/she attains a stable 7 to 9 m/s descent rate. The Glider has no tendency to front rosette or become pitch unstable. To exit the B stall the pilot releases the B slowly until the glider has regained its shape and then the **last 15 cm fast** to prevent the glider from entering deep stall. The KARMA / KARMA SP can be controlled directionally in the B stall by pulling more on one B riser than on the other to create a turn in any direction. The B-stall is a safe controlled way of losing altitude fast without any forward speed.

15.7 BIG EARS

Altitude can be lost in a controlled way by collapsing both tips. To do this, take the outermost A-line (attached on its own riser) on both side and pull them down until the tips collapse. Pulling one side at a time may be more comfortable and easier, especially for smaller pilots. This should close about 30% of the wing in total. It is possible to steer with weight shift.

To increase the sink rate the pilot can push the speed system after he/she has collapsed the tips. This can give up to about 7 m/s sink-rate with about 38-km/h forward speed. To exit, release the speed system and then release the tip A-lines.

It may sometimes be necessary to apply a little brake to open the tips. If using the brakes to open the tips, it is best to open one tip at a time, this avoids reducing your air-speed.

CAUTION:

DO NOT ATTEMPT ANY EXTREME MANEUVRES WITH THE TIPS COLLAPSED AS THIS DOUBLES THE LOAD ON THE CENTER LINES AND ATTACHMENT POINTS AND COULD LEAD TO FAILURE.





15.8 DEEP STALL OR PARACHUTAL STALL

Under normal flying conditions the KARMA / KARMA SP will have no tendency to enter deep stall. All gliders can however under certain conditions enter and stay in deep stall configuration (as a result of ageing of materials, improper maintenance or pilot induced).

15.8.1 Signs of parachutal stall

- The pilot has very little or no forwards speed and no wind in his face.
- The glider will be fully open but the cells will be bulging in and not out on the bottom surface.
- The glider might have a very slow turning sensation.
- You will have an increased vertical descent.

15.8.2 Exit from parachutal stall

It is important to recognize this situation. Most accidents involving parachutal stall happen because the pilot did not realize that he was in deep stall.

The best way to exit a parachutal stall is to pull all the A risers down to get the wing flying again. The pilot can pull the riser down until the wing starts to fly again. The moment the wing starts to fly the pilot should release the A riser, or the wing might suffer a frontal collapse.

Alternatively the pilot can push the speed bar to lower the angle of attack and get the wing flying again.

By pulling one or both brakes while in deep stall the pilot can accidentally enter a full stall or spin.

15.9 SPIRAL DIVES

The KARMA / KARMA SP have very good behaviour in spiral and have no tendency to stick in the spiral. By progressively applying brake on one side the glider can be put into a spiral dive. Safe high sink rates can be achieved like this. The spiral has to be exited slowly by releasing the brake over one complete turn or the glider may pitch forward and possibly suffer a collapse.

Care must be taken that the pilot has enough height to exit the spiral safely.

Sink rates in excess of 19m/s can be obtained.

CAUTION:

SOME GLIDERS CAN BE NEUTRAL IN SPIRAL AND MAY NOT EXIT WITHOUT PILOT INPUT. TO EXIT A NEUTRAL SPIRAL THE PILOT HAS TO LEAN HIS/HER WEIGHT TO THE OUTSIDE OF THE TURN OR APPLY BRAKE ON THE OUTSIDE WING. AS SOON AS THE GLIDER STARTS TO SLOW DOWN IN THE SPIRAL THE OUTSIDE BRAKE MUST BE RELEASED.

PILOTS CAN SUFFER BLACK OUTS IN SPIRALS AND THE PILOT HAS TO EXIT THE SPIRAL AS SOON AS he/she FEELS ANY ABNORMAL SYMPTOMS (Black dots in field of vision or light-headedness).



15.10 STRONG TURBULENCE

NEVER FLY IN STRONG TURBULENCE!

If you unexpectedly encounter strong turbulence, fly with about 20% brake applied to increase the internal pressure and the angle of attack of the canopy and land as soon as possible. If the air is turbulent on landing approach, land with Big Ears.

Learn to fly actively and to anticipate collapses and prevent them by applying brake when needed before you have unwanted collapses.

15.11 STEERING NOT FUNCTIONING

If the pilot cannot reach the brake or steering lines for any reason or if they are not functioning properly, (for example: If they break on a damaged point) he or she can control the glider by pulling down on the rear risers.

Care must be taken when steering like this, as much less input is needed to turn the wing and the response of the wing is also much slower than when using the brakes.

IF YOU PULL TOO MUCH ON ONE OR BOTH RISERS THE GLIDER WILL SPIN OR STALL.

On the landing flare the pilot should be especially careful not to stall the glider too high.

16 LANDING

Before landing the pilot should determine the wind direction, usually by checking a windsock, flags, smoke or your drift over the ground while doing one or more 360° turns.

- Always land into the wind.
- At a height of about 50 meters your landing setup should begin. The most commonly used one is to head into the wind and depending on the wind strength the pilot should reach his/her landing point by making s-turns.
- At a height of about 15 meters the final part of your descent should be made at trim speed into the wind.
- At a height between half a meter and one meter you can gently flair the glider by pulling gradually down on the brakes to the stall point. When top-landing it is sometimes not necessary to flare or a much smaller flare may be required, especially in strong ridge conditions.

16.1 TREE LANDING

If it is not possible to land in an open area, steer into the wind towards an unobstructed tree and do a normal landing approach as if the tree is your landing spot. Flare as for a normal landing. On impact hold your legs together and protect your face with your arms.

After any tree landing it is very important to check all the lines, line measurements, and the canopy for damage.



16.2 WATER LANDING

As you approach landing, release all the buckles (and cross-bracing if present) of the harness except for one leg. Just before landing, release the remaining buckle. It is advisable to enter the water downwind. Let the canopy rotate completely forward until it hits the water with the leading edge openings; the air inside will then be trapped, forming a big air mattress and giving the pilot more time to escape. Less water will enter the canopy this way, making the recovery much easier. Get away from the glider and lines as soon as possible, to avoid entanglement. Remember that a ballast bag can be emptied and then inflated with air for a flotation aid.

The canopy should be carefully inspected after a water landing, since it is very easy to cause internal damage to the ribs if the canopy is lifted while containing water. Always lift the canopy by the trailing edge, not by the lines or top or bottom surface fabric.

16.3 LANDING IN TURBULENCE

One of the safest ways to land a glider in turbulent conditions is to use Big Ears. This reduces the chances of getting a collapse while on final approach. Use weight shift to control your approach. It is possible to keep the ears in until you are ready to flare the glider. Simply release the A-risers and flare the glider, starting a little higher than usual. Practice this in normal conditions before you need it in an emergency.

17 PACKING

Spread the canopy completely out on the ground. Separate the lines to the left and the right side of the glider. If the risers are removed from the harness, join the two risers together by passing one carabineers loop through the other. This keeps them neatly together and helps to stop line tangles.

Fold the canopy alternately from the right and left sides, working towards the center, press out the air, working from the rear towards the front. Place the risers at the trailing edge of the folded canopy and use them to finally roll up the canopy.

18 MAINTENANCE & CLEANING

Cleaning should be carried out with water and if necessary, gentle soap. If the glider comes in contact with salt water, clean thoroughly with fresh water. Do not use solvents of any kind, as this may remove the protective coatings and destroy the fabric.

18.1 Butt holes (Velcro closure on trailing edge tip)

In order to empty sand and small stones from the glider simply shake the sand or small stones into the wing tip and open the **Butt holes** (**Velcro closure** on trailing edge tip) to empty. Do not forget to close the **Butt holes** afterwards.





19 STORAGE

When the glider is not in use, the glider should be stored in a cool, dry place. A wet glider should first be dried (out of direct sunlight). Protect the glider against sunlight (UV radiation). When on the hill keep the glider covered or in the bag. Never store or transport the glider near paint, petrol or any other chemicals.

Do not leave your paraglider in the trunk of a car or exposed to the sun.

Temperatures on a hot summer's day in a closed environment: car, etc. can easily reach over 60°C

At these temperatures Nylon permanently changes its characteristics which may alter the behavior and shape of the wing.

It will cause permanent damage to the paraglider, rendering it non-airworthy. APCO's warranty will not be applicable.

20 DAMAGE

Using spinnaker repair tape (for non-siliconized cloth) can repair tears in the sail (up to 5cm). A professional repairer should repair greater damage.

21 GENERAL ADVICE

A qualified person or agent of the company should check the glider every year.

The glider is carefully manufactured and checked by the factory. Never make changes to the canopy or the lines. Changes can introduce dangerous flying characteristics and will not improve flying performance.

Do not put the glider in direct sunlight when not necessary. In order to protect the glider during transportation or waiting time we recommend one of our lightweight storage bags.

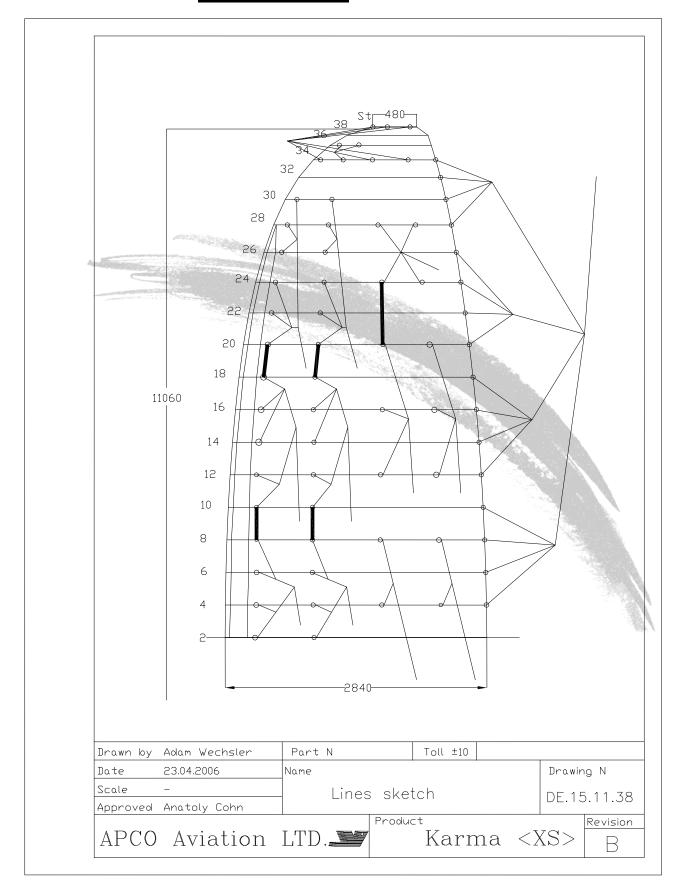
If you have any doubts about flying conditions-do not begin.

If you have any questions, please contact your dealer or us.

Lastly, be equipped with a certified emergency parachute and helmet on every flight.

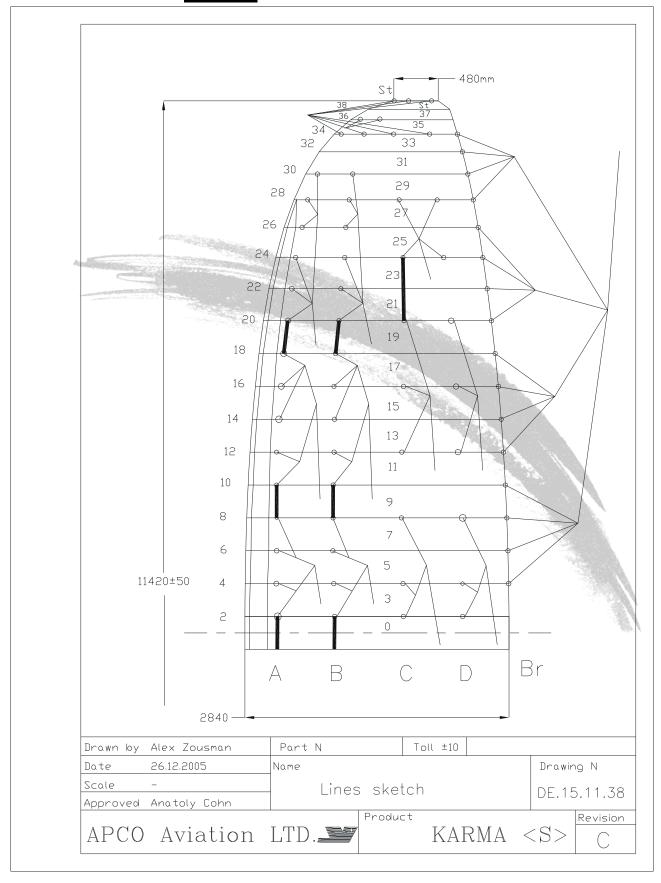


22 KARMA EXTRA SMALL SKETCHES





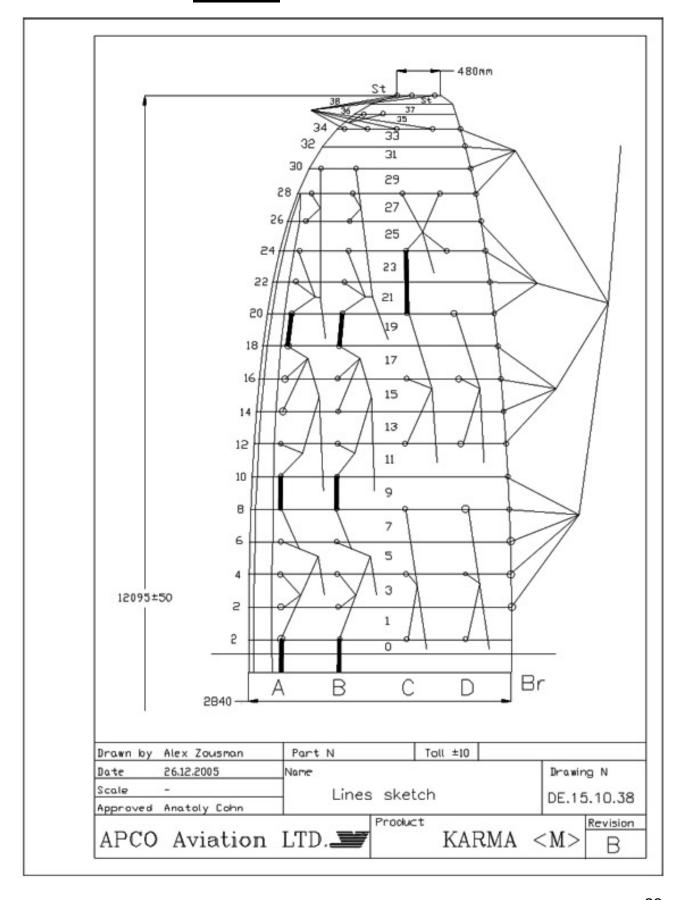
23 KARMA SMALL SKETCHES





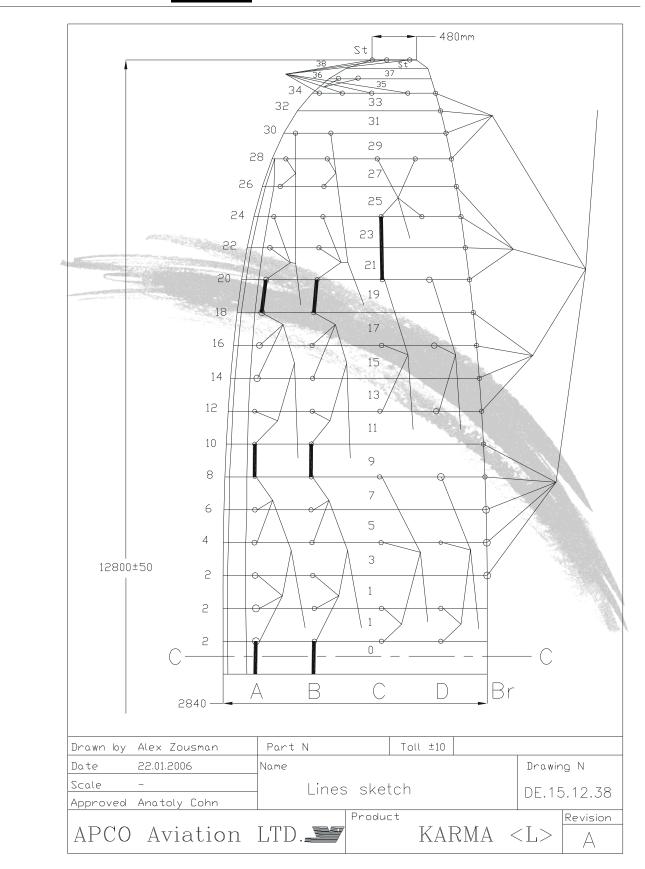
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24 KARMA <u>MEDIUM</u> SKETCHES





25 KARMA LARGE SKETCHES







26 KARMA EXTRA SMALL CERTIFICATION

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TEST REPORT DHV 03 APCO K	ARMA XS				
	Type Apco Karma XS				
Certific	Certificate-No DHV GS-01-1536-06				
Holder of cer	rtificate Apco Aviation Ltd.				
Manuf	acturer Apco Aviation Ltd.				
Classi	fication 1 GH				
Win	nch tow Yes				
Number of seats min / Number of	of seats 1 / 1 max				
Accel	erator? Yes				
Trin	mmers? No				
	BEHAVIOUR AT MIN WEIGHT IN FLIGHT(55 KG)	BEHAVIOUR AT MAX WEIGHT IN FLIGHT(75 KG)			
Take off	1	1			
Inflation	evenly, immediately	evenly, immediately			
Rising behaviour	immediately comes over pilot	immediately comes over pilot			
Take off speed	average	average			
Take off handling	easy	easy			
Straight flight	1	1			
Roll damping	average	average			
Turn handling	1	1			
Spin tendency	slight	slight			
Control travel		high			
Agility	average	average			
Symmetric stall	1	1			
Deep-stall limit	late > 75 cm	late > 75 cm			
Full stall limit	late > 80 cm	late > 80 cm			
Increase in steering power	high	high			
Front collapse	1	1			
Pre-acceleration	slight	slight			
Opening behaviour	spontaneous, quickly	spontaneous, quickly			





Asymmetric collapse	1	1
Turn tendency	< 90 degrees	< 90 degrees
Change of course	_	90 - 180 degrees
Rate of turn	_	slight
Rate of turn	with deceleration	Siigitt
Max. roll/pitch angle		less than 45 degrees
Loss of altitude		slight
Stabilization		spontaneous
Opening behaviour	spontaneous	spontaneous
Countersteering an asymmetric collapse	1	1
Stabilization	countersteering easy	countersteering easy
Control travel	high	high
Control pressure increase	high	high
Turn in opposite direction	easy, no tendency to stall	easy, no tendency to stall
Opening behaviour	spontaneous, quickly	spontaneous, quickly
Full stall, symm. exit	1	1
Spin out of straight flight	1	1
Spin out of turn	1	1
Spiral dive	1	1
Entry	easy	easy
Spin tendency	slight	slight
Exit	turn continues through < 180 degrees	turn continues through < 180 degrees
Sink rate after 720 ° [m/s]	8	10
B-line stall	1	1
Entry	easy	easy
Exit	spontaneous	spontaneous
Big ears	1	1
Entry		easy
Recovery	spontaneous, quickly	spontaneous, quickly
Landing	1	1
Landing behaviour	easy	easy





Front collapse (accelerated)	1	1
Pre-acceleration	slight	slight
Opening behaviour	spontaneous, delayed	spontaneous, delayed
Asymmetric collapse (accelerated)	1	1
Turn tendency	< 90 degrees	< 90 degrees
Change of course	90 - 180 degrees	90 - 180 degrees
Rate of turn	slight	slight
	with deceleration	
Max. roll/pitch angle	less than 45 degrees	less than 45 degrees
Loss of altitude	slight	slight
Stabilization	spontaneous	spontaneous
Opening behaviour	spontaneous	spontaneous
Big ears accelerated	1	1
Entry	easy	easy
Recovery	spontaneous, quickly	spontaneous, quickly
Supplementary remarks		







27 KARMA SMALL CERTIFICATION

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TEST REPORT DHV 03 APCO K	ARMA S		
	Type Apco KARMA S		
Certific	cate-No DHV GS-01-1537-06		
Holder of cer	rtificate Apco Aviation Ltd.		
Manuf	acturer Apco Aviation Ltd.		
Classi	fication 1 GH		
Win	nch tow Yes		
Number of seats min / Number of	of seats 1 / 1 max		
Accel	erator? Yes		
Trir	mmers? No	_	
	BEHAVIOUR AT MIN WEIGHT IN FLIGHT(60 KG)	BEHAVIOUR AT MAX WEIGHT IN FLIGHT(90 KG)	
Take off	1	1	
Inflation	evenly, immediately	evenly, immediately	
Rising behaviour	immediately comes over pilot	immediately comes over pilot	
Take off speed	average	average	
Take off handling	easy	easy	
Straight flight	1	1	
Roll damping	average	average	
Turn handling	1	1	
Spin tendency	slight	slight	
Control travel	high	high	
Agility	average	average	
Symmetric stall	1	1	
Deep-stall limit	late > 75 cm	late > 75 cm	
Full stall limit	late > 80 cm	late > 80 cm	
Increase in steering power	high	high	
Front collapse	1	1	
Pre-acceleration	slight	slight	
Opening behaviour	spontaneous, delayed	spontaneous, quickly	





Asymmetric collapse	1	1
Turn tendency	< 90 degrees	< 90 degrees
Change of course	_	90 - 180 degrees
Rate of turn	_	slight
	3	3
Max. roll/pitch angle	less than 45 degrees	less than 45 degrees
Loss of altitude		slight
Stabilization		spontaneous
Opening behaviour	-	spontaneous
Countersteering an asymmetric collapse	1	1
Stabilization	countersteering easy	countersteering easy
Control travel	high	high
Control pressure increase	high	high
Turn in opposite direction	easy, no tendency to stall	easy, no tendency to stall
Opening behaviour	spontaneous, quickly	spontaneous, quickly
Full stall, symm. exit	1	1
Spin out of straight flight	1	1
Spin out of turn	1	1
Spiral dive	1	1
Entry	easy	easy
Spin tendency	slight	slight
Exit	turn continues through < 180 degrees	turn continues through < 180 degrees
Sink rate after 720 ° [m/s]	10	10
B-line stall	1	1
Entry	easy	easy
Exit	spontaneous	spontaneous
Big ears	1	1
Entry	easy	easy
Recovery	spontaneous, quickly	spontaneous, quickly
Landing	1	1
Landing behaviour	easy	easy





Front collapse (accelerated)	1	1
Pre-acceleration	slight	slight
Opening behaviour	spontaneous, delayed	spontaneous, delayed
Asymmetric collapse (accelerated)	1	1
Turn tendency	< 90 degrees	< 90 degrees
Change of course	90 - 180 degrees	90 - 180 degrees
Rate of turn	slight	slight
Max. roll/pitch angle	less than 45 degrees	less than 45 degrees
Loss of altitude	slight	slight
Stabilization	spontaneous	spontaneous
Opening behaviour	spontaneous	spontaneous
Big ears accelerated	1	1
Entry	easy	easy
Recovery	spontaneous, quickly	spontaneous, quickly
Supplementary remarks		







28 KARMA <u>MEDIUM</u> CERTIFICATION

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TEST REPORT DHV 03 APCO KARMA M			
Type Apco KARMA M			
Certific	cate-No DHV GS-01-1538-06		
Holder of cer	rtificate Apco Aviation Ltd.		
Manuf	acturer Apco Aviation Ltd.		
Classi	fication 1 GH		
Wir	nch tow Yes		
Number of seats min / Number of	of seats 1 / 1 max		
Accel	erator? Yes		
Trir	mmers? No		
	BEHAVIOUR AT MIN WEIGHT IN FLIGHT(80 KG)	BEHAVIOUR AT MAX WEIGHT IN FLIGHT(110 KG)	
Take off	1	1	
Inflation	evenly, immediately	evenly, immediately	
Rising behaviour	immediately comes over pilot	immediately comes over pilot	
Take off speed	average	average	
Take off handling	easy	easy	
Straight flight	1	1	
Roll damping	average	average	
Turn handling	1	1	
Spin tendency	slight	slight	
Control travel	high	high	
Agility	average	average	
Symmetric stall	1	1	
Deep-stall limit	late > 75 cm	late > 75 cm	
Full stall limit	late > 80 cm	late > 80 cm	
Increase in steering power	high	high	
Front collapse	1	1	
Pre-acceleration	slight	slight	
Opening behaviour	spontaneous, delayed	spontaneous, quickly	





Asymmetric collapse	1	1
Turn tendency	-	< 90 degrees
Change of course		90 - 180 degrees
Rate of turn	slight	slight
Max. roll/pitch angle		less than 45 degrees
Loss of altitude	_	slight
Stabilization		spontaneous
Opening behaviour	spontaneous	spontaneous
Countersteering an asymmetric collapse	1	1
Stabilization	countersteering easy	countersteering easy
Control travel	high	high
Control pressure increase	high	high
Turn in opposite direction	easy, no tendency to stall	easy, no tendency to stall
Opening behaviour	spontaneous, quickly	spontaneous, quickly
Full stall, symm. exit	1	1
Spin out of straight flight	1	1
Spin out of turn	1	1
Spiral dive	1	1
Entry	easy	easy
Spin tendency	slight	slight
Exit	turn continues through < 180 degrees	turn continues through < 180 degrees
Sink rate after 720 ° [m/s]	10	10
B-line stall	1	1
Entry	easy	easy
Exit	spontaneous	spontaneous
Big ears	1	1
Entry		easy
Recovery	spontaneous, quickly	spontaneous, quickly
Landing	1	1
Landing behaviour	easy	easy





Front collapse (accelerated)	1	1
Pre-acceleration	slight	slight
Opening behaviour	spontaneous, delayed	spontaneous, delayed
Asymmetric collapse (accelerated)	1	1
Turn tendency	< 90 degrees	< 90 degrees
Change of course	90 - 180 degrees	90 - 180 degrees
Rate of turn	slight	slight
Max. roll/pitch angle	less than 45 degrees	less than 45 degrees
Loss of altitude	slight	slight
Stabilization	spontaneous	spontaneous
Opening behaviour	spontaneous	spontaneous
Big ears accelerated	1	1
Entry	easy	easy
Recovery	spontaneous, quickly	spontaneous, quickly
Supplementary remarks		







29 KARMA LARGE CERTIFICATION

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TEST REPORT DHV 03 APCO KARMA L

Type Apco Karma L

Certificate-No DHV GS-01-1539-06

Holder of certificate Apco Aviation Ltd.

Manufacturer Apco Aviation Ltd.

Classification 1 GH

Winch tow Yes

Number of seats min / Number of seats 1 / 1

Accelerator? Yes

Trimmers? No		
	BEHAVIOUR AT MIN WEIGHT IN FLIGHT(100 KG)	BEHAVIOUR AT MAX WEIGHT IN FLIGHT(130 KG)
Take off	1	1
Inflation	evenly, immediately	evenly, immediately
Rising behaviour	immediately comes over pilot	immediately comes over pilot
Take off speed	average	average
Take off handling	easy	easy
Straight flight	1	1
Roll damping	average	average
Turn handling	1	1
Spin tendency	slight	slight
Control travel	high	high
Agility	average	average
Symmetric stall	1	1
Deep-stall limit	late > 75 cm	late > 75 cm
Full stall limit	late > 80 cm	late > 80 cm
Increase in steering power	high	high





Front collapse	1	1
Pre-acceleration	slight	slight
Opening behaviour	spontaneous, delayed	spontaneous, quickly
Asymmetric collapse	1	1
Turn tendency	< 90 degrees	< 90 degrees
Change of course	90 - 180 degrees	90 - 180 degrees
Rate of turn	slight	slight
Max. roll/pitch angle	less than 45 degrees	less than 45 degrees
Loss of altitude	slight	slight
Stabilization	spontaneous	spontaneous
Opening behaviour	spontaneous	spontaneous
Countersteering an asymmetric collapse	1	1
Stabilization	countersteering easy	countersteering easy
Control travel	high	high
Control pressure increase	high	high
Turn in opposite direction	easy, no tendency to stall	easy, no tendency to stall
Opening behaviour	spontaneous, quickly	spontaneous, quickly
Full stall, symm. exit	1	1
Spin out of straight flight	1	1
Spin out of turn	1	1
Spiral dive	1	1
Entry	easy	easy
Spin tendency	slight	slight
Exit	turn continues through < 180 degrees	turn continues through < 180 degrees
Sink rate after 720 ° [m/s]	10	10
B-line stall	1	1
Entry	easy	easy
Exit	spontaneous	spontaneous
Big ears	1	1
Entry	easy	easy

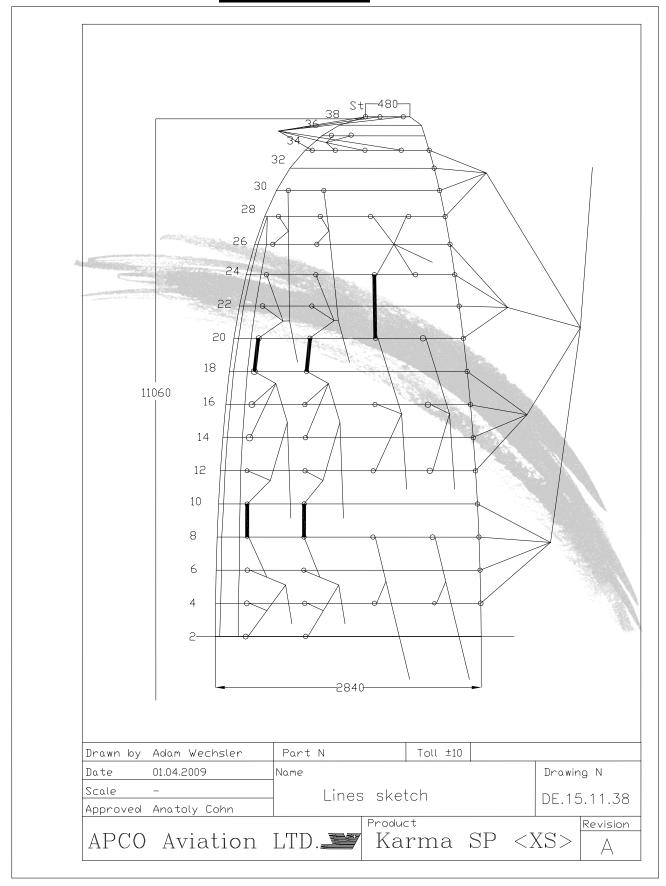


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Recovery	spontaneous, quickly	spontaneous, quickly
Landing	1	1
Landing behaviour	easy	easy
Front collapse (accelerated)	1	1
Pre-acceleration	slight	slight
Opening behaviour	spontaneous, delayed	spontaneous, delayed
Asymmetric collapse (accelerated)	1	1
Turn tendency	< 90 degrees	< 90 degrees
Change of course	90 - 180 degrees	90 - 180 degrees
Rate of turn	slight	slight
Max. roll/pitch angle Loss of altitude Stabilization Opening behaviour	slight spontaneous	less than 45 degrees slight spontaneous spontaneous
Big ears accelerated	1	1
Entry		easy spontaneous, quickly
	spontaneous, quickly	sportaneous, quickly
Supplementary remarks		· · · · · · · · · · · · · · · · · · ·

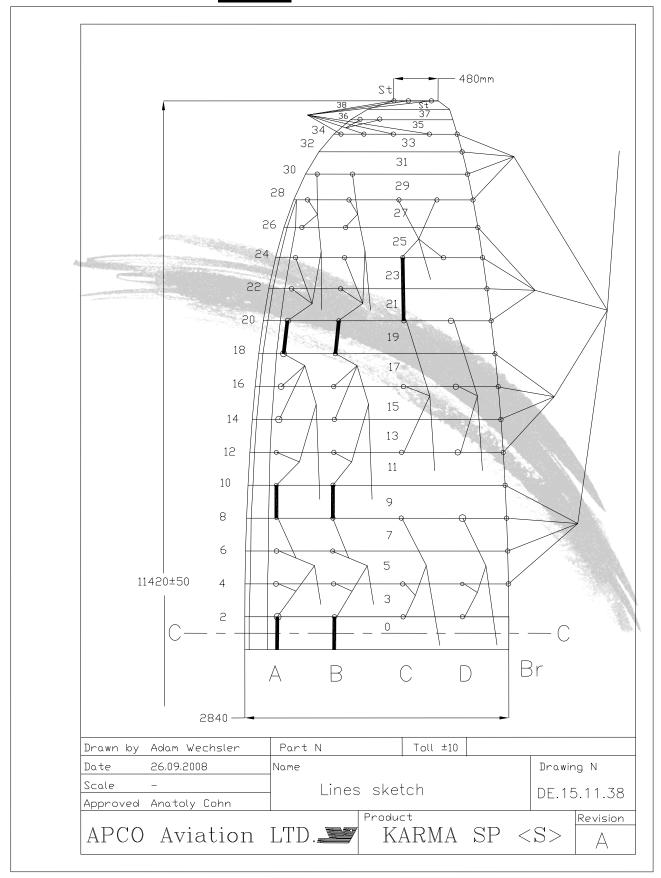


30 KARMA SP EXTRA SMALL SKETCHES



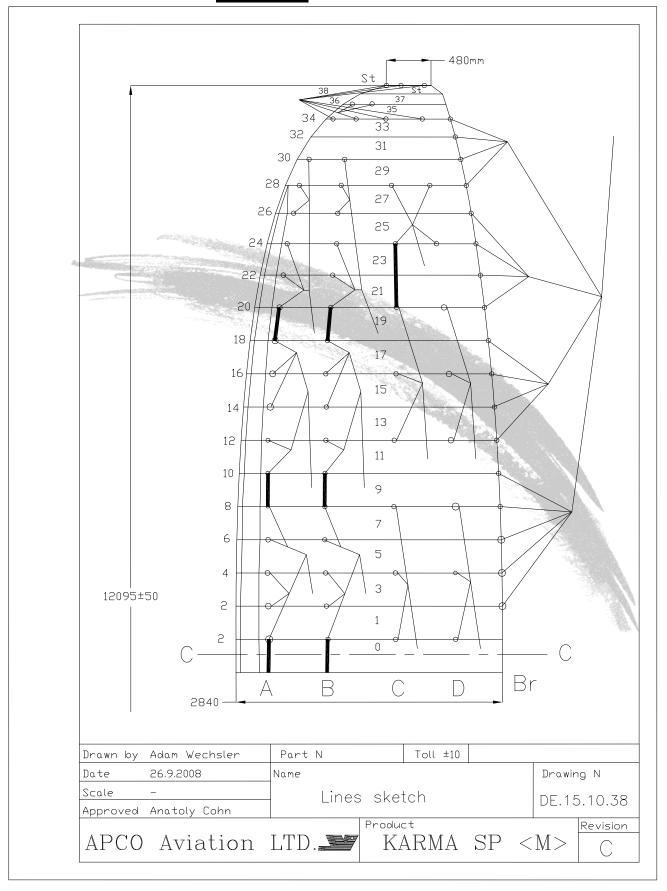


31 KARMA SP SMALL SKETCHES



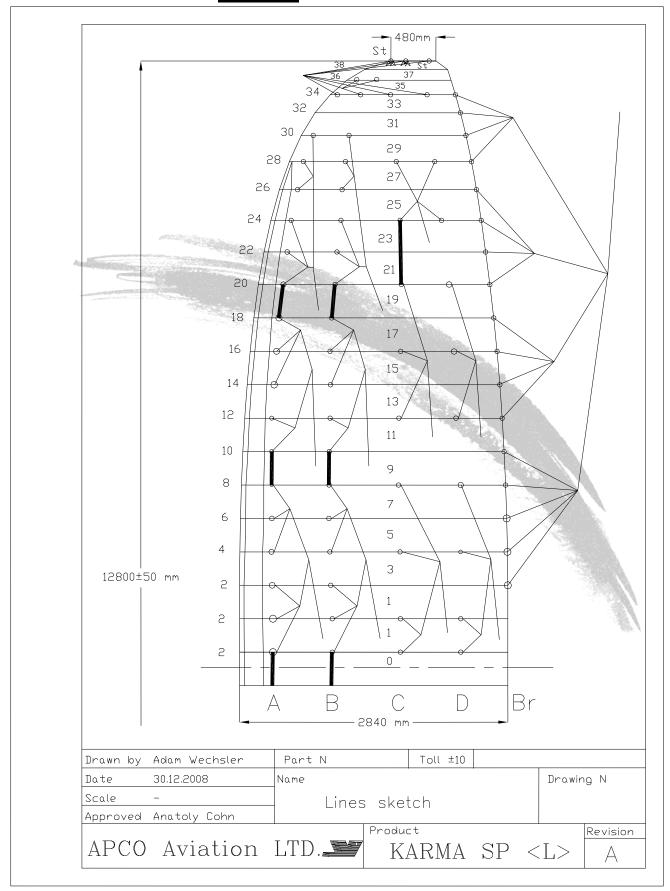


32 KARMA SP MEDIUM SKETCHES





33 KARMA SP <u>LARGE</u> SKETCHES







34 KARMA SP SMALL CERTIFICATION

para-test.com



Air Turquoise SA Rte du Pré-au-Comte 8 | CH-1844 Villeneuve tel. •41 21 965 65 65 | mobile •41 79 202 52 30 info@para-test.com

Apco Aviation Ltd.

Mr Cohn Anatoly 7, Chalamish St., Industrial park 38900 Caesarea Israel

Certificate

The hereunder sample of paraglider has been tested in accordance with the following standards: EN 926-2:2005 & EN 926-1:2006



Certification number	PG_0183.2008
Manufacturer	Apco Aviation Ltd.
Glider model	Karma Sport S
Category	В
Maximum weight in flight (kg)	90 kg
Minimum weight in flight (kg)	60 kg
Glider's weight (kg)	5.4 kg

Date of flight test

Flight tests,	07. 10. 2008
Serial number	PR00413

Best Regards,

Alain Zoller

Randi Eriksen



para-test.com



Air Turquoise SA Rte du Pré-au-Comte B | CH-1844 Villeneuve tel. +41 21 965 65 65 | mobile +41 79 202 52 30 info@para-test.com

Flight test report



Manufacturer Apco Aviation Ltd. Certification number PG_0183.2008

Address 7, Chalamish St., Industrial park Date of flight test 07. 10. 2008

38900 Caesarea

Israel

Representative none Place of test Villeneuve

Glider model Karma Sport S Classification B

Trimmer no

Test pilot Dupont Philippe Thurnheer Claude Harness Advance - Light Gin - Genie III

Total weight in flight (kg) 60 90

rotal weight in hight (kg)	00		30	
1. Inflation/Take-off	A			
Rising behaviour	Smooth, easy and constant rising	Α	Smooth, easy and constant rising	Α
Special take off technique required	No	Α	No	Α
2. Landing	A			
Special landing technique required	No	Α	No	Α
3. Speed in straight flight	Α			
Trim speed more than 30 km/h	Yes	Α	Yes	Α
Speed range using the controls larger than 10 km/h	Yes	Α	Yes	Α
Minimum speed	Less than 25 km/h	Α	Less than 25 km/h	Α
4. Control movement	A			
Max. weight in flight up to 80 kg				
Symmetric control pressure / travel	Increasing / greater than 55 cm	Α	not available	0
Max. weight in flight 80 kg to 100 kg				
Symmetric control pressure / travel	not available	0	Increasing / greater than 60 cm	Α
Max. weight in flight greater than 100 kg				
Symmetric control pressure / travel	not available	0	not available	0
5. Pitch stability exiting accelerated flight	A			
Dive forward angle on exit	Dive forward less than 30°	Α	Dive forward less than 30°	Α
Collapse occurs	No	Α	No	Α
6. Pitch stability operating controls during accelerated flight	Α			
Collapse occurs	No	Α	No	Α
7. Roll stability and damping	A			
Oscillations	Reducing	Α	Reducing	Α
8. Stability in gentle spirals	A			
Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
9. Behaviour in a steeply banked turn	В			
Sink rate after two turns	Up to 12 m/s	Α	More than 14 m/s	В
10. Symmetric front collapse	A			
Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit / Change of course	Dive forward 0° to 30° / Keeping course	Α	Dive forward 0° to 30° / Keeping course	Α
Cascade occurs	No	Α	No	Α
With accelerator				
Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α

Flight test report: PG_0183.2008 / page 1 of 3



www.apcoaviation.com				3
Dive forward angle on exit / Change of course	Dive forward 0° to 30° / Keeping course	Α	Dive forward 0° to 30° / Keeping course	A
Cascade occurs	No	Α	No	Α
11. Exiting deep stall (parachutal stall)	A			
Deep stall achieved	Yes	Α	Yes	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
Change of course	Changing course less than 45°	Α	Changing course less than 45°	Α
Cascade occurs	No	Α	No	Α
12. High angle of attack recovery	A			
Recovery	not available	0	Spontaneous in less than 3 s	Α
Cascade occurs	not available	0	No	Α
13. Recovery from a developed full stall	Α			
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
Collapse	No collapse	Α	No collapse	Α
Cascade occurs (other than collapses)	No	Α	No	Α
Rocking back	Less than 45°		Less than 45°	Α
Line tension	Most lines tight	Α	Most lines tight	Α
14. Asymmetric collapse	A			
With 50% collapse				
Change of course until re-inflation / Maximum dive forward or roll angle	Less than 90° / Dive or roll angle 0° to 15°	Α	Less than 90° / Dive or roll angle 0° to 15°	Α
Re-inflation behaviour	Spontaneous re-inflation	Α	Spontaneous re-inflation	Α
Total change of course	Less than 360°	Α	Less than 360°	Α
Collapse on the opposite side occurs	No	Α	No	Α
Twist occurs	No	Α	No	Α
Cascade occurs	No	Α	No	Α
With 75% collapse				
Change of course until re-inflation / Maximum dive forward or roll angle	Less than 90° / Dive or roll angle 0° to 15°	Α	Less than 90° / Dive or roll angle 15° to 45°	Α
Re-inflation behaviour	Spontaneous re-inflation	Α		Α
Total change of course	Less than 360°	Α	Less than 360°	Α
Collapse on the opposite side occurs	No	Α	No	Α
Twist occurs	No	Α	No	Α
Cascade occurs	No	Α	No	Α
With 50% collapse and accelerator				
Change of course until re-inflation / Maximum dive forward or roll angle	0° to 15°		Less than 90° / Dive or roll angle 15° to 45°	Α
Re-inflation behaviour	Spontaneous re-inflation	A	Spontaneous re-inflation	A
Total change of course	Less than 360°	Α	Less than 360°	A
Collapse on the opposite side occurs	No	Α	No	A
Twist occurs	No	A	No	A
Cascade occurs	No	А	No	Α
With 75% collapse and accelerator Change of course until re-inflation / Maximum dive forward or	Less than 90° / Dive or roll angle	Α		Α
roll angle Re-inflation behaviour	0° to 15° Spontaneous re-inflation	Α	15° to 45° Spontaneous re-inflation	Α
Total change of course	Less than 360°	A	Less than 360°	A
Collapse on the opposite side occurs	No	A	No	A
Twist occurs	No	A	No	A
Cascade occurs	No	A	No	A
15. Directional control with a maintained asymmetric collapse	A			
Able to keep course	Yes	Α	Yes	Α
180° turn away from the collapsed side possible in 10 s	Yes		Yes	A
Amount of control range between turn and stall or spin	More than 50 % of the symmetric control travel		More than 50 % of the symmetric control travel	A
16. Trim speed spin tendency	A			
Spin occurs	No	Α	No	Α

Flight test report: PG_0183.2008 / page 2 of 3



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17. Low speed spin tendency	Α			
Spin occurs	No	Α	No	Α
18. Recovery from a developed spin	A			
Spin rotation angle after release	Stops spinning in less than 90°	Α	Stops spinning in less than 90°	Α
Cascade occurs	No	Α	No	Α
19. B-line stall	Α			
Change of course before release	not available	0	Changing course less than 45°	Α
Behaviour before release	not available	0	Remains stable with straight span	Α
Recovery	not available	0	Spontaneous in less than 3 s	Α
Dive forward angle on exit	not available	0	Dive forward 0° to 30°	Α
Cascade occurs	not available	0	No	Α
20. Big ears	В			
Entry procedure	Dedicated controls	Α	Dedicated controls	Α
Behaviour during big ears	Stable flight	Α	Stable flight	Α
Recovery	Spontaneous in 3 s to 5 s	В	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
21. Big ears in accelerated flight	В			
Entry procedure	Dedicated controls	Α	Dedicated controls	Α
Behaviour during big ears	Stable flight	Α	Stable flight	Α
Recovery	Recovery through pilot action in less than a further 3 s	В	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
Behaviour immediately after releasing the accelerator while maintaining big ears	Stable flight	Α	Stable flight	Α
22. Behaviour exiting a steep spiral	A			
Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
Turn angle to recover normal flight	Less than 720°, spontaneous recovery	Α	Less than 720°, spontaneous recovery	Α
Sink rate when evaluating spiral stability [m/s]	14		17	
23. Alternative means of directional control	Α			
180° turn achievable in 20 s	Yes	Α	Yes	Α
Stall or spin occurs	No	Α	No	Α
24. Any other flight procedure and/or configuration described in the user's manual	0			
Procedure works as described	not available	0	not available	0
Procedure suitable for novice pilots	not available	0	not available	0
Cascade occurs	not available	0	not available	0
25. Comments of test pilot				
Comments	Impossible to make a B-line stall.			

Flight test report: PG_0183.2008 / page 3 of 3





35 KARMA SP MEDIUM CERTIFICATION

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Air Turquoise SA Rte du Pré-au-Comte 8 | CH-1844 Villeneuve tel. •41 21 965 65 65 | mobile •41 79 202 52 30 info@para-test.com

Apco Aviation Ltd. Mr Cohn Anatoly 7, Chalamish St., Industrial park 38900 Caesarea Israel

Certificate

The hereunder sample of paraglider has been tested in accordance with the following standards: EN 926-2:2005 & EN 926-1:2006



 Certification number
 PG_0184.2008

 Manufacturer
 Apco Aviation Ltd.

 Glider model
 Karma Sport M

 Category
 B

 Maximum weight in flight (kg)
 110 kg

 Minimum weight in flight (kg)
 80 kg

 Glider's weight (kg)
 5.8 kg

Date of flight test

 Flight tests
 23. 09. 2008

 Serial number
 140290

Best Regards,

Alain Zoller

Randi Eriksen



para-test.com



Air Turquoise SA Rte du Pré-au-Comte B | CH-1844 Villeneuve tel. +41 21 965 65 65 | mobile +41 79 202 52 30 info@para-test.com

Flight test report



PG_0184.2008 Manufacturer Apco Aviation Ltd. Certification number Address 7, Chalamish St., Industrial park Date of flight test 23.09.2008

38900 Caesarea

Representative Place of test Villeneuve None

Glider model Karma Sport M Classification

Trimmer

Test pilot Thurnheer Claude Zoller Alain

Harness Sky paragliders - Axel II M Sol Paragliders - Slider L

Total weight in flight (kg)	80		110	
1. Inflation/Take-off	A			
Rising behaviour	Smooth, easy and constant rising	Α	Smooth, easy and constant rising	Α
Special take off technique required	No	Α	No	Α
2. Landing	A			
Special landing technique required	No	Α	No	Α
3. Speed in straight flight	A			
Trim speed more than 30 km/h	Yes	Α	Yes	Α
Speed range using the controls larger than 10 km/h	Yes	Α	Yes	Α
Minimum speed	Less than 25 km/h	Α	Less than 25 km/h	Α
4. Control movement	A			
Max. weight in flight up to 80 kg				
Symmetric control pressure / travel	Increasing / greater than 55 cm	Α	not available	0
Max. weight in flight 80 kg to 100 kg				
Symmetric control pressure / travel	not available	0	not available	0
Max. weight in flight greater than 100 kg				
Symmetric control pressure / travel	not available	0	Increasing / greater than 65 cm	Α
5. Pitch stability exiting accelerated flight	A			
Dive forward angle on exit	Dive forward less than 30°	Α	Dive forward less than 30°	Α
Collapse occurs	No	Α	No	Α
6. Pitch stability operating controls during accelerated flight	Α			
Collapse occurs	No	Α	No	Α
7. Roll stability and damping	A			
Oscillations	Reducing	Α	Reducing	Α
8. Stability in gentle spirals	A			
Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
9. Behaviour in a steeply banked turn	В			
Sink rate after two turns	12 m/s to 14 m/s	Α	More than 14 m/s	В
10. Symmetric front collapse	A			
Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit / Change of course	Dive forward 0° to 30° / Keeping course	Α	Dive forward 0° to 30° / Keeping course	Α
Cascade occurs	No	Α	No	Α
With accelerator				
Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α

Flight test report: PG_0184.2008 / page 1 of 3

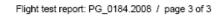


<u>www.apcoaviation.com</u>				
Dive forward angle on exit / Change of course	Dive forward 0° to 30° / Keeping course	Α	Dive forward 0° to 30° / Keeping course	Α
Cascade occurs	No	Α	No	Α
11. Exiting deep stall (parachutal stall)	A			
Deep stall achieved	Yes	Α	Yes	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
Change of course	Changing course less than 45°	Α	Changing course less than 45°	Α
Cascade occurs	No	Α	No	Α
12. High angle of attack recovery	Α			
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Cascade occurs	No	Α	No	Α
13. Recovery from a developed full stall	Α			
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
Collapse	No collapse	Α	No collapse	Α
Cascade occurs (other than collapses)	No .	Α	No	Α
Rocking back	Less than 45°	Α	Less than 45°	Α
Line tension	Most lines tight	Α	Most lines tight	Α
14. Asymmetric collapse	В			
With 50% collapse				
Change of course until re-inflation / Maximum dive forward or roll angle	Less than 90° / Dive or roll angle 0° to 15°	Α	Less than 90° / Dive or roll angle 0° to 15°	Α
Re-inflation behaviour	Spontaneous re-inflation	Α	Spontaneous re-inflation	Α
Total change of course	Less than 360°	Α	Less than 360°	Α
Collapse on the opposite side occurs	No	Α	No	Α
Twist occurs	No	Α	No	Α
Cascade occurs	No	Α	No	Α
With 75% collapse				
Change of course until re-inflation / Maximum dive forward or roll angle	Less than 90° / Dive or roll angle 15° to 45°	Α	Less than 90° / Dive or roll angle 15° to 45°	Α
Re-inflation behaviour	Spontaneous re-inflation	Α	Spontaneous re-inflation	Α
Total change of course	Less than 360°	Α	Less than 360°	Α
Collapse on the opposite side occurs	No	Α	No	Α
Twist occurs	No	Α	No	Α
Cascade occurs	No	Α	No	Α
With 50% collapse and accelerator				
Change of course until re-inflation / Maximum dive forward or roll angle	Less than 90° / Dive or roll angle 15° to 45°	Α	Less than 90° / Dive or roll angle 15° to 45°	Α
Re-inflation behaviour	Spontaneous re-inflation	Α	Spontaneous re-inflation	Α
Total change of course	Less than 360°	Α	Less than 360°	Α
Collapse on the opposite side occurs	No	Α	No	Α
Twist occurs	No	Α	No	Α
Cascade occurs	No	Α	No	Α
With 75% collapse and accelerator				
Change of course until re-inflation / Maximum dive forward or roll angle	Less than 90° / Dive or roll angle 15° to 45°	Α	90° to 180° / Dive or roll angle 15° to 45°	В
Re-inflation behaviour	Spontaneous re-inflation	Α	Spontaneous re-inflation	Α
Total change of course	Less than 360°	Α	Less than 360°	Α
Collapse on the opposite side occurs	No	Α	No	Α
Twist occurs	No	Α	No	Α
Cascade occurs	No	Α	No	Α
15. Directional control with a maintained asymmetric collapse	Α			
Able to keep course	Yes	Α	Yes	Α
180° turn away from the collapsed side possible in 10 s	Yes	Α	Yes	Α
Amount of control range between turn and stall or spin	More than 50 % of the symmetric control travel	Α	More than 50 % of the symmetric control travel	Α
16. Trim speed spin tendency	A			

Flight test report: PG_0184.2008 / page 2 of 3



www.apcoaviation.com_				
17. Low speed spin tendency	A			
Spin occurs	No	Α	No	Α
18. Recovery from a developed spin	Α			
Spin rotation angle after release	Stops spinning in less than 90°	Α	Stops spinning in less than 90°	Α
Cascade occurs	No	Α	No	Α
19. B-line stall	Α			
Change of course before release	Changing course less than 45°	Α	Changing course less than 45°	Α
Behaviour before release	Remains stable with straight span	Α	Remains stable with straight span	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
Cascade occurs	No	Α	No	Α
20. Big ears	Α			
Entry procedure	Dedicated controls	Α	Dedicated controls	Α
Behaviour during big ears	Stable flight	Α	Stable flight	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
21. Big ears in accelerated flight	Α			
Entry procedure	Dedicated controls	Α	Dedicated controls	Α
Behaviour during big ears	Stable flight	Α	Stable flight	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
Behaviour immediately after releasing the accelerator while maintaining big ears	Stable flight	Α	Stable flight	Α
22. Behaviour exiting a steep spiral	A			
Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
Turn angle to recover normal flight	Less than 720°, spontaneous recovery	Α	Less than 720°, spontaneous recovery	Α
Sink rate when evaluating spiral stability [m/s]	15		22	
23. Alternative means of directional control	Α			
180° turn achievable in 20 s	Yes	Α	Yes	Α
Stall or spin occurs	No	Α	No	Α
24. Any other flight procedure and/or configuration described in the user's manual	0			
Procedure works as described	not available	0	not available	0
Procedure suitable for novice pilots	not available	0	not available	0
Cascade occurs	not available	0	not available	0
25. Comments of test pilot				
Comments				





36 KARMA SP LARGE CERTIFICATION





Air Turquoise SA Rte du Pré-au-Comte 8 | CH-1844 Villeneuve tel. •41 21 965 65 65 | mobile •41 79 202 52 30 info@para-test.com

Apco Aviation Ltd. Mr Cohn Anatoly 7, Chalamish St., Industrial park 38900 Caesarea Israel

Certificate

The hereunder sample of paraglider has been tested in accordance with the following standards: EN 926-2:2005 & EN 926-1:2006



 Certification number
 PG_0231.2009

 Manufacturer
 Apco Aviation Ltd.

 Glider model
 Karma Sport L

 Category
 B

 Maximum weight in flight (kg)
 130 kg

 Minimum weight in flight (kg)
 100 kg

 Glider's weight (kg)
 6.4 kg

Date of flight test

 Flight tests
 12. 03. 2009

 Serial number
 140778

 Load test
 21. 03. 2009

 Serial number
 140778

Best Regards, Alain Zoller

Randi Eriksen



para-test.com



Air Turquoise SA Rte du Pré-au-Comte 8 | CH-1844 Villeneuve tel. +41 21 965 65 65 | mobile +41 79 202 52 30 info@para-test.com

Flight test report



Manufacturer Apco Aviation Ltd. Certification number PG_0231.2009

Address 7, Chalamish St., Industrial park Date of flight test 12. 03. 2009

38900 Caesarea 12. 03. 2009

Israel

Representative None Place of test Annecy
Glider model Karma Sport L Classification B

Trimmer no

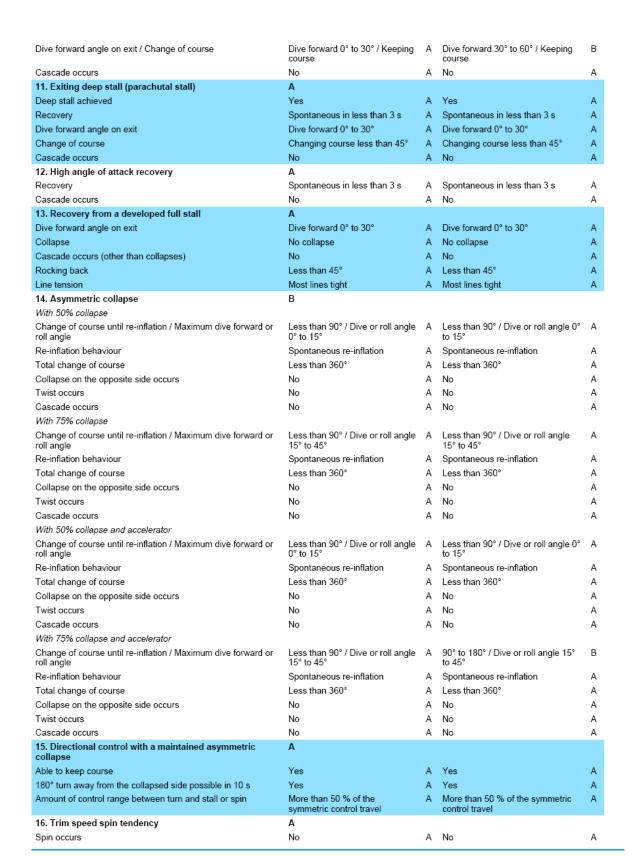
Test pilot Thurnheer Claude Zoller Alain

Harness Sup'Air - Evo XC M Gin Gliders - Gingo 2 L

Total weight in flight (kg	100		130	
1. Inflation/Take-off	Α			
Rising behaviour	Smooth, easy and constant rising	Α	Smooth, easy and constant rising	Α
Special take off technique required	No	Α	No	Α
2. Landing	Α			
Special landing technique required	No	Α	No	Α
3. Speed in straight flight	Α			
Trim speed more than 30 km/h	Yes	Α	Yes	Α
Speed range using the controls larger than 10 km/h	Yes	Α	Yes	Α
Minimum speed	Less than 25 km/h	Α	Less than 25 km/h	Α
4. Control movement	Α			
Max. weight in flight up to 80 kg				
Symmetric control pressure / travel	not available	0	not available	0
Max. weight in flight 80 kg to 100 kg				
Symmetric control pressure / travel	not available	0	not available	0
Max. weight in flight greater than 100 kg				
Symmetric control pressure / travel	Increasing / greater than 65 cm	Α	Increasing / greater than 65 cm	Α
5. Pitch stability exiting accelerated flight	Α			
Dive forward angle on exit	Dive forward less than 30°	Α	Dive forward less than 30°	Α
Collapse occurs	No	Α	No	Α
6. Pitch stability operating controls during accelerated flight	Α			
Collapse occurs	No	Α	No	Α
7. Roll stability and damping	Α			
Oscillations	Reducing	Α	Reducing	Α
8. Stability in gentle spirals	Α			
Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
9. Behaviour in a steeply banked turn	A			
Sink rate after two turns	12 m/s to 14 m/s	Α	Up to 12 m/s	Α
10. Symmetric front collapse	В			
Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit / Change of course	Dive forward 0° to 30° / Keeping course	Α	Dive forward 0° to 30° / Keeping course	Α
Cascade occurs	No	Α	No	Α
With accelerator				
Entry	Rocking back less than 45°	Α	Rocking back less than 45°	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α

Flight test report: PG_0231.2009 / page 1 of 3



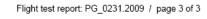


Flight test report: PG_0231.2009 / page 2 of 3





17. Low speed spin tendency	Α			
Spin occurs	No	Α	No	Α
18. Recovery from a developed spin	Α			
Spin rotation angle after release	Stops spinning in less than 90°	Α	Stops spinning in less than 90°	Α
Cascade occurs	No	Α	No	Α
19. B-line stall	Α			
Change of course before release	Changing course less than 45°	Α	Changing course less than 45°	Α
Behaviour before release	Remains stable with straight span	Α	Remains stable with straight span	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
Cascade occurs	No	Α	No	Α
20. Big ears	Α			
Entry procedure	Dedicated controls	Α	Dedicated controls	Α
Behaviour during big ears	Stable flight	Α	Stable flight	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
21. Big ears in accelerated flight	Α			
Entry procedure	Dedicated controls	Α	Dedicated controls	Α
Behaviour during big ears	Stable flight	Α	Stable flight	Α
Recovery	Spontaneous in less than 3 s	Α	Spontaneous in less than 3 s	Α
Dive forward angle on exit	Dive forward 0° to 30°	Α	Dive forward 0° to 30°	Α
Behaviour immediately after releasing the accelerator while maintaining big ears	Stable flight	Α	Stable flight	Α
22. Behaviour exiting a steep spiral	Α			
Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α
Turn angle to recover normal flight	Less than 720°, spontaneous recovery	Α	Less than 720°, spontaneous recovery	Α
Sink rate when evaluating spiral stability [m/s]	18		14	
23. Alternative means of directional control	Α			
180° turn achievable in 20 s	Yes	Α	Yes	Α
Stall or spin occurs	No	Α	No	Α
24. Any other flight procedure and/or configuration described in the user's manual	0			
Procedure works as described	not available	0	not available	0
Procedure suitable for novice pilots	not available	0	not available	0
Cascade occurs	not available	0	not available	0
25. Comments of test pilot				
Comments				

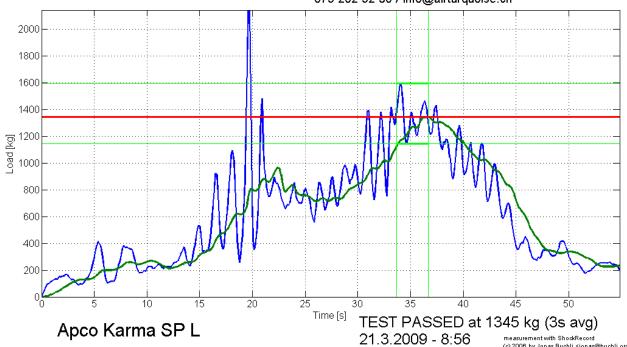




37 LOAD TEST

Air Turquoise Homologations LOAD DIAGRAM

Alain Zoller Rue de la Poterlaz 6, CP-10 CH-1844 Villeneuve 079 202 52 30 / info@airturquoise.ch









APCO wishes you many hours of enjoyable flying.

Take Air!

